Engineering Software

Copyright © 1996

P.O. Box 2134 Kensington, MD 20891 Phone: (301) 919-9670 E-Mail: info@engineering-4e.com http://www.engineering-4e.com

Power Cycles Efficiency Derivation

Here engineering students and professionals get familiar with the ideal simple and basic power cycles and their T - s and p - V diagrams, operation, cycle efficiency derivation and major performance trends when air is considered as the working fluid.

Performance Objectives:

Introduce basic energy conversion engineering assumptions and equations

Know basic elements of Carnot Cycle, Brayton Cycle, Otto Cycle and Diesel Cycle and their T - s and p - V diagrams

Be familiar with Carnot Cycle, Brayton Cycle, Otto Cycle and Diesel Cycle operation

Provide cycle efficiency derivation for Carnot Cycle, Brayton Cycle, Otto Cycle and Diesel Cycle

Understand general Carnot Cycle, Brayton Cycle, Otto Cycle and Diesel Cycle performance trends

enoitymuzzA gnineenignE

The energy conversion analysis presented in this webinar considers ideal (isentropic) operation and air, argon, helium and nitrogen are considered as the working fluid. Furthermore, the following assumptions are valid:

Power Cycles

Single species consideration -- fuel mass flow rate is ignored and its impact on the properties of the working fluid

Basic equations hold (continuity, momentum and energy equations) Specific heat is constant

Power Cycle Components/Processes

Single species consideration Basic equations hold (continuity, momentum and energy equations) Specific heat is constant Basic Engineering Equations

Basic Conservation Equations

Continuity Equation m = ρvA [kg/s]

Momentum Equation F = (vm + pA)_{out - in} [N]

Energy Equation Q - W = $((h + v^2/2 + gh)m)_{out - in}$ [kW]

Basic Engineering Equations

Ideal Gas State Equation pv = RT [kJ/kg]

Perfect Gas $c_p = constant [kJ/kg^K]$

> Kappa X = c_p/c_v [/]

For air: $\chi = 1.4$ [/], R = 0.2867 [kJ/kg*K] and $c_p = 1.004$ [kJ/kg*K]



Carnot Cycle Schematic Layout



Entropy -- s [kJ/kg*K]

Carnot Cycle T - s Diagram

The thermal cycle efficiency can be given as a function of specific external work (specific net power output) and heat added to the working fluid as follows:

$$\eta = w/q_h = (w_t - w_c)/q_h = (q_h - q_l)/q_h = 1 - q_l/q_h = 1 - T_1\Delta s/T_2\Delta s = 1 - T_1/T_2 = 1 - T_R/T_A$$

where

- η thermal efficiency [/]
- w specific external work (specific net power output) [kJ/kg]
- w_t expansion specific power output [kJ/kg]
- w_c compression specific power input [kJ/kg]
- q_h heat added to the working fluid [kJ/kg]

q_I - heat rejected from the working fluid [kJ/kg]

Δs - entropy change during heat addition and heat rejection [kJ/kg*K]

T_A - temperature during heat addition [K]

T_R - temperature during heat rejection [K]

For isentropic compression and expansion:

$$T_2/T_1 = (p_2/p_1)^{(\varkappa-1)/\varkappa} = T_3/T_4 = (p_3/p_4)^{(\varkappa-1)/\varkappa}$$

and

$$\kappa = c_p/c_v$$
 - for air $\kappa = 1.4$ [/]

p₁, p₂, p₃, p₄ - pressure values at points 1, 2, 3 and 4 [atm]

T₁, T₂, T₃, T₄ - temperature values at points 1, 2, 3 and 4 [K]

Again, it follows that

 $\eta = 1 - T_1/T_2 = 1 - T_R/T_A$

The Carnot Cycle efficiency is not dependent on the working fluid properties.

Governing Equations

 $T_2/T_1 = (p_2/p_1)^{(\varkappa-1)/\varkappa}$

 $T_3/T_4 = (p_3/p_4)^{(\varkappa-1)/\varkappa}$

 $\kappa = c_p/c_v$

pv = RT

 $q_h = T_2 \Delta s = T_A \Delta s$

 $q_I = T_1 \Delta s = T_R \Delta s$

 $\eta = 1 - T_1/T_2 = 1 - T_R/T_A$

Carnot Cycle Efficiency



Compressor Inlet Temperature: 298 [K]

Carnot Cycle Efficiency



Turbine Inlet Temperature: 800 [K]

Brayton Cycle (Gas Turbine)



Working Fluid In

Working Fluid Out

Brayton Cycle (Gas Turbine) Schematic Layout -- Open Cycle

Brayton Cycle (Gas Turbine)



Brayton Cycle Schematic Layout -- Closed Cycle



Entropy -- s [kJ/kg*K]

Brayton Cycle (Gas Turbine) T - s Diagram

The thermal cycle efficiency can be given as a function of specific external work (specific net power output) and heat added to the working fluid as follows:

$$\eta = w/q_h = (w_t - w_c)/q_h = (q_h - q_l)/q_h$$

or

$$\eta = 1 - q_l/q_h = 1 - (c_p(T_4 - T_1))/(c_p(T_3 - T_2)) = 1 - (T_1(T_4/T_1 - 1))/(T_2(T_3/T_2 - 1))$$

where

- η thermal efficiency [/]
- w specific external work (specific net power output) [kJ/kg]
- w_t expansion specific power output [kJ/kg]
- w_c compression specific power input [kJ/kg]
- W external work (net power output) [kW]

- W_t expansion power output [kW]
- W_c compression power input [kW]
- q_h heat added to the working fluid [kJ/kg]
- q_I heat rejected from the working fluid [kJ/kg]
- c_p specific heat at constant pressure [kJ/kg*K]
- c_v specific heat at constant volume [kJ/kg*K]
- m working fluid mass flow rate [kg/s]
- r_p compression ratio [/]

For isentropic compression and expansion:

 $T_2/T_1 = (p_2/p_1)^{(\varkappa-1)/\varkappa}$

 $T_3/T_4 = (p_3/p_4)^{(\varkappa-1)/\varkappa}$

Brayton Cycle (Gas Turbine)

Knowing that

 $p_3/p_4 = p_2/p_1$

where

 $\kappa = c_p/c_v$ - for air $\kappa = 1.4$ [/]

 p_1 , p_2 , p_3 , p_4 - pressure values at points 1, 2, 3 and 4 [atm] T₁, T₂, T₃, T₄ - temperature values at points 1, 2, 3 and 4 [K] It follows that

 $T_3/T_4 = T_2/T_1$

or

 $T_3/T_2 = T_4/T_1$

Therefore, after some mathematical operations the thermal efficiency is:

 $\eta = 1 - T_1/T_2 = 1 - T_4/T_3$

If the temperature ratio is substituted in terms of the compression ratio:

 $\eta = 1 - 1/r_{p}^{(\varkappa - 1)/\varkappa}$

where

 $r_{p} = p_{2}/p_{1}$

Governing Equations

 $T_2/T_1 = (p_2/p_1)^{(\varkappa-1/)\varkappa}$

- $p_2/p_1 = (T_2/T_1)^{\kappa/(\kappa-1)}$
- $T_3/T_4 = (p_3/p_4)^{(\varkappa-1/)\varkappa}$
- $p_3/p_4 = (T_3/T_4)^{\varkappa/(\varkappa-1)}$

 $\varkappa = c_p/c_v$

pv = RT

 $w = q_h - q_l$

 $q_h = c_p(T_3 - T_2)$

 $q_{I} = c_{p}(T_{4} - T_{1})$

Governing Equations (Continued)

$$w = c_{p}(T_{3} - T_{2}) - c_{p}(T_{4} - T_{1})$$
$$W = (c_{p}(T_{3} - T_{2}) - c_{p}(T_{4} - T_{1}))m$$
$$\eta = 1 - 1/r_{p}^{(\varkappa - 1)/\varkappa}$$

 $r_{p} = p_{2}/p_{1}$

Brayton Cycle (Gas Turbine) Efficiency



Working Fluid: Air



Brayton Cycle (Gas Turbine) Specific Power Output



Brayton Cycle (Gas Turbine) Power Output



Working Fluid: Air



Volume -- V [m^3]

Otto Cycle p - V Diagram





Otto Cycle T - s Diagram

The thermal cycle efficiency can be given as a function of specific external work (specific net power output) and heat added to the working fluid as follows:

$$\eta = w/q_h = (w_e - w_c)/q_h = (q_h - q_l)/q_h$$

or

$$\eta = 1 - q_l/q_h = 1 - (c_v(T_4 - T_1))/(c_v(T_3 - T_2)) = 1 - (T_1(T_4/T_1 - 1))/(T_2(T_3/T_2 - 1))$$

where

- η thermal efficiency [/]
- w specific external work (specific net power output) [kJ/kg]
- we expansion specific power output [kJ/kg]

w_c - compression specific power input [kJ/kg]

W - external work (net power output) [kW]

W_e - expansion power output [kW]

W_c - compression power input [kW]

q_h - heat added to the working fluid [kJ/kg]

q_I - heat rejected from the working fluid [kJ/kg]

c_p - specific heat at constant pressure [kJ/kg*K]

c_v - specific heat at constant volume [kJ/kg*K]

m - working fluid mass flow rate [kg/s]

 ϵ - compression ratio [/]

For isentropic compression and expansion:

$$T_2/T_1 = (p_2/p_1)^{(\varkappa-1)/\varkappa} = (V_1/V_2)^{(\varkappa-1)/\varkappa}$$

 $\mathsf{T}_4/\mathsf{T}_3 = (\mathsf{p}_4/\mathsf{p}_3)^{(\varkappa\cdot 1)/\varkappa} = (\mathsf{V}_3/\mathsf{V}_4)^{(\varkappa\cdot 1)}$

Knowing that

 $V_3/V_4 = V_2/V_1$

where

$$\kappa = c_p/c_v$$
 - for air $\kappa = 1.4$ [/]

 V_1 , V_2 , V_3 , V_4 - volume values at points 1, 2, 3 and 4 [m³]

p₁, p₂, p₃, p₄ - pressure values at points 1, 2, 3 and 4 [atm]

T₁, T₂, T₃, T₄ - temperature values at points 1, 2, 3 and 4 [K]

It follows that

$$T_3/T_4 = T_2/T_1 = (V_1/V_2)^{(\varkappa-1)} = \epsilon^{(\varkappa-1)}$$

where

 $\varepsilon = V_1/V_2$

Therefore, after some mathematical operations the thermal efficiency is:

$$\eta = 1 - T_1/T_2 = 1 - 1/(V_1/V_2)^{(\varkappa-1)}$$

If the temperature ratio is substituted in terms of the volume/compression ratio:

 $\eta = 1 - 1/\epsilon^{(\varkappa-1)}$

Governing Equations

 $T_2/T_1 = (V_1/V_2)^{(\varkappa-1)}$ $V_1/V_2 = (T_2/T_1)^{1/(\varkappa-1)}$ $T_3/T_4 = (V_4/V_3)^{(\varkappa-1)}$ $V_4/V_3 = (T_3/T_4)^{1/(\varkappa-1)}$ $\kappa = c_p/c_v$ pv = RT $w = q_h - q_l$ $q_{h} = c_{v}(T_{3} - T_{2})$ $q_{I} = c_{v}(T_{4} - T_{1})$

Governing Equations (Continued)

$$w = c_v(T_3 - T_2) - c_v(T_4 - T_1)$$
$$W = (c_v(T_3 - T_2) - c_v(T_4 - T_1))m$$
$$\eta = 1 - 1/\epsilon^{(\varkappa - 1)}$$

 $\epsilon = V_1/V_2$

Otto Cycle Efficiency



→ V1/V2 [/]

Working Fluid: Air

Otto Cycle Power Output



Working Fluid: Air

Ambient Temperature: 298 [K] -- Number of Revolutions: 60 [1/s]

For a Given Geometry of a Four Cylinder and Four Stroke Otto Engine





Volume -- V [m^3]

Diesel Cycle p - V Diagram



Entropy -- s [kJ/kg*K]

Diesel Cycle T - s Diagram

The thermal cycle efficiency can be given as a function of specific external work (specific net power output) and heat added to the working fluid as follows:

$$\eta = w/q_h = (w_e - w_c)/q_h = (q_h - q_l)/q_h$$

or

$$\eta = 1 - q_l/q_h = 1 - (c_v(T_4 - T_1))/(c_p(T_3 - T_2)) = 1 - (T_1(T_4/T_1 - 1))/(\kappa T_2(T_3/T_2 - 1))$$

where

- η thermal efficiency [/]
- w specific external work (specific net power output) [kJ/kg]
- w_e expansion specific power output [kJ/kg]

- w_c compression specific power input [kJ/kg]
- W external work (net power output) [kW]
- W_e expansion power output [kW]
- W_c compression power input [kW]
- q_h heat added to the working fluid [kJ/kg]
- q_I heat rejected from the working fluid [kJ/kg]
- c_p specific heat at constant pressure [kJ/kg*K]
- c_v specific heat at constant volume [kJ/kg*K]
- m working fluid mass flow rate [kg/s]
- ϵ compression ratio [/]

 ϕ - cut off ratio [/]

For isentropic compression and expansion:

 $T_2/T_1 = (p_2/p_1)^{(\varkappa-1)/\varkappa} = (V_1/V_2)^{(\varkappa-1)}$ $T_4/T_3 = (p_4/p_3)^{(\varkappa-1)/\varkappa} = (V_3/V_4)^{(\varkappa-1)}$

where

$$\kappa = c_p/c_v$$
 - for air $\kappa = 1.4$ [/]

 V_1 , V_2 , V_3 , V_4 - volume values at points 1, 2, 3 and 4 [m³]

p₁, p₂, p₃, p₄ - pressure values at points 1, 2, 3 and 4 [atm]

T₁, T₂, T₃, T₄ - temperature values at points 1, 2, 3 and 4 [K]

Knowing that

 $s_3 - s_2 = s_4 - s_1$

and

 $s_3 - s_2 = c_p ln(T_3/T_2)$

 $s_4 - s_1 = c_v ln(T_4/T_1)$

 $s_1,\,s_2,\,s_3,\,s_4$ - specific entropy values at points 1, 2, 3 and 4 [kJ/kg*K]

It follows

 $(T_3/T_2)^{\varkappa} = T_4/T_1$

It follows that

 $T_3/T_4 = T_2/T_1 = (V_1/V_2)^{(\varkappa-1)} = \epsilon^{(\varkappa-1)}$

When combustion takes place at a constant pressure:

 $T_3/T_2 = V_3/V_2$

where

 $\varepsilon = V_1/V_2$

 $\varphi = V_3/V_2$

Therefore, after some mathematical operations the thermal efficiency is:

$$\eta = 1 - (T_1((T_3/T_2)^{\varkappa} - 1)))/(\varkappa T_2(T_3/T_2 - 1))$$

If the temperature ratio is substituted in terms of the volume/compression ratio:

 $η = 1 - (φ^{\varkappa} - 1)/(\varkappa ε^{(\varkappa - 1)}(φ - 1))$

Governing Equations

 $T_2/T_1 = (V_1/V_2)^{(\varkappa-1)}$ $V_1/V_2 = (T_2/T_1)^{1/(\varkappa-1)}$ $T_3/T_4 = (V_4/V_3)^{(\varkappa-1)}$ $V_4/V_3 = (T_3/T_4)^{1/(\varkappa-1)}$ $\kappa = c_p/c_v$ pv = RT $w = q_h - q_l$ $q_h = c_p(T_3 - T_2)$ $q_1 = c_v(T_4 - T_1)$

Governing Equations (Continued)

$$w = c_{p}(T_{3} - T_{2}) - c_{v}(T_{4} - T_{1})$$

$$W = (c_{p}(T_{3} - T_{2}) - c_{v}(T_{4} - T_{1}))m$$

$$\eta = 1 - (\phi^{\varkappa} - 1)/(\varkappa \epsilon^{(\varkappa - 1)}(\phi - 1))$$

$$\epsilon = V_{1}/V_{2}$$

$$\phi = V_{3}/V_{2}$$

Diesel Cycle Efficiency



Diesel Cycle Efficiency



Ambient Temperature: 298 [K]

Diesel Cycle Cut Off Ratio



Diesel Cycle Power Output



Working Fluid: Air

Ambient Temperature: 298 [K] -- Number of Revolutions: 60 [1/s]

For a Given Geometry of a Four Cylinder and Four Stroke Diesel Engine