

Engineering Software

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Power and Propulsion Systems Analysis 1.1 - Tutorial

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Program Description

Engineering Software has developed a new Windows based software package, **Power and Propulsion Systems Analysis**, that quickly and reliably analyzes power cycles, power cycle components/processes and compressible flow.

This software package should prove to be a good tool for those who are involved at various levels with design, operation and management of power and propulsion systems. It should provide you with the opportunity to more quickly and effectively do your work, explore more options, save time and give more confidence in carrying out your calculations.

To get a free evaluation copy of the program, place an order, find out more about how you can profit or benefit from this software package, please send an e-mail to **info@engineering-4e.com** or call **(301) 540-3605**.

Power Cycles

Carnot

Brayton (Power and Propulsion)

Rankine

Power Cycle Components/Processes

Compression

Combustion (Coal/Oil/Gas)

Expansion

Compressible Flow

Mach Number

Nozzle

Normal Shock

Diffuser

Thrust

Hardware Requirements and Software Compatibility

80486 or higher microprocessor

16 MB RAM

10 MB available on hard drive

IBM compatible systems:

Microsoft® Windows 98®, Windows 2000®, Windows ME®,

Windows XP® and Windows Vista®

Note: Required Microsoft® Office 2007® (Microsoft® Access 2007®) for Windows Vista operating system!

Distributed on CD ROM

Free Technical and Product Support

30 Day Money Back Guarantee

Trademarks: All product names, company names and service marks are trademarks or registered trademarks of their respective companies.

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Claim Sheet

Engineering Software products allow quick and reliable calculation of thermodynamic and transport properties of gaseous, liquid and solid species, contain coefficients for the calculation of physical properties, steam approximations for both saturated and superheated areas, analyze power cycles, power cycle components/processes and compressible flow.

The aforementioned engineering calculations are valid under the following assumptions:

Thermodynamic and Transport Properties

Single species consideration

Ideal gas approach is used ($pV=RT$)

Specific heat is not constant

Coefficients describing thermodynamic and transport properties were obtained from the NASA Glenn Research Center at Lewis Field in Cleveland, OH -- such coefficients conform with the standard reference temperature of 298.15 K (77 F) and the JANAF tables

Power Cycles

Single species consideration -- fuel mass flow rate ignored and its impact on the properties of the working fluid

Basic equations hold (continuity, momentum and energy equations)

Specific heat is constant

Power Cycle Components/Processes

Single species consideration

Basic equations hold (continuity, momentum and energy equations)

Specific heat is constant

Compressible Flow

Single species consideration

Basic equations hold (continuity, momentum and energy equations)

Specific heat is constant

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How to Use the Program

In each section, subsection of the **Power and Propulsion Systems Analysis** program, the user needs to change one or more input values in order to calculate a new case. Input values are in boxes with white background and can be changed by clicking on each individual box or even by using the arrow keys and changing the current value. Output values cannot be modified, changed by the user and they are in boxes with black background.

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Limited Warranty

This software package is sold **AS IS**, without warranty as to its performance. The entire risk (as to) the quality and of the performance of this computer software program is assumed by the user.

However, to the original purchaser only, **Engineering Software** warrants the medium on which the program is recorded to be free from defects in materials and faulty workmanship under normal use and service for a period of thirty (30) days from the date of purchase. If during this period a defect on the medium should occur, the medium may be returned to **Engineering Software** or to an authorized **Engineering Software** distributor, and **Engineering Software** will replace the medium without charge to you. Your sole and exclusive remedy in the event of a defect is expressly limited to replacement of the medium as provided above.

If the failure of the medium, in the judgment of **Engineering Software** resulted from accident, abuse, or misapplication of the medium, then **Engineering Software** shall have no responsibility to replace the medium under the terms of this warranty.

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Suggestion/Evaluation Form

Please FAX or Mail This Suggestion/Evaluation Form To:

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Name:

Title:

Company/Organization:

Street Address:

City:

State:

Zip:

Phone:

FAX:

E-Mail:

Power and Propulsion Systems Analysis

Quality:

Usefulness:

Possible areas of improvement:

Other suggestions:

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License Agreement

This software is the property of **Engineering Software** and is protected by federal copyright law. While **Engineering Software** continues to own the software, you will have certain rights to use the software after your acceptance of this license. Your rights and obligations with respect to the use of this software are as follows:

You may:

- use one copy of the software on a single computer,
- make one copy of the software for backup purposes, and
- use the software on a network, provided that you have a licensed copy of the software for each computer that can access the software over the network

You may not:

- copy the documentation which accompanies the software,
- sublicense, rent or lease any portion of the software, and
- reverse engineer, decompile, disassemble, modify, translate, make any attempt to discover the source code of the software, or create derivative works from the software

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Power Cycles

This section provides analysis of a few power cycles (Carnot, Brayton and Rankine).

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Power Cycles: Carnot

This subsection provides analysis of the Carnot cycle.

Input Values:

Heat Addition Temperature, Heat Rejection Temperature

Output Values:

Cycle Efficiency, Heat Rate

Assumptions:

Isentropic compression and expansion. Heat addition and rejection occur at constant temperature. Specific heat is constant.

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Running - Power Cycles: Carnot

STEP 1

Enter the desired heat addition temperature value or go with the default temperature value.

STEP 2

Enter the desired heat rejection temperature value or go with the default temperature value.

STEP 3

Click on the Calculate button to start the calculation of the Carnot cycle output values for the chosen input values.

STEP 4

When done with Steps 1 through 3, click on the Exit button to go back to the Main menu.

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Power Cycles: Brayton

This subsection provides analysis of the Brayton cycle for both power generation and propulsion applications.

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Power Cycles: Brayton: Power (Ideal)

This subsection provides analysis of the Brayton cycle for the power generation application.

Input Values:

Working Fluid, Working Fluid Mass Flow Rate, Compressor Inlet Temperature, Compressor Inlet Pressure, Turbine Inlet Temperature, Turbine Inlet Pressure, Fuel HHV

Output Values:

Power Output, Fuel Consumption, Cycle Efficiency, Heat Rate

Assumptions:

Isentropic compression and expansion. Ideal combustion, heat transfer. Fuel mass flow rate is ignored when calculating the gas turbine power output. No pressure loss. Specific heat is constant.

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Running - Power Cycles: Brayton: Power (Ideal)

STEP 1

Select the desired working fluid or go with the default working fluid.

STEP 2

Enter the desired compressor inlet temperature value or go with the default temperature value.

STEP 3

Enter the desired compressor inlet pressure value or go with the default pressure value.

STEP 4

Enter the desired turbine inlet temperature value or go with the default temperature value.

STEP 5

Enter the desired turbine inlet pressure value or go with the default turbine inlet pressure value.

STEP 6

Enter the desired mass flow rate value or go with the default mass flow rate value.

STEP 7

Enter the desired fuel HHV value or go with the default fuel HHV value.

STEP 8

Click on the Calculate button to start the calculation of the Brayton cycle output values for the chosen input values.

STEP 9

When done with Steps 1 through 8, click on the Exit button to go back to the Main menu.

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Power Cycles: Brayton: Propulsion (Ideal)

This subsection provides analysis of the Brayton cycle for the propulsion application.

Input Values:

Working Fluid, Working Fluid Mass Flow Rate, Compressor Inlet Temperature, Compressor Inlet Pressure, Turbine Inlet Temperature, Turbine Inlet Pressure, Fuel HHV

Output Values:

Thrust, Fuel Consumption

Assumptions:

Isentropic compression and expansion. Ideal combustion, heat transfer. Fuel mass flow rate is ignored when calculating the gas turbine power output. Ambient pressure is equal to compressor inlet pressure. No pressure loss. Specific heat is constant.

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Running - Power Cycles: Brayton: Propulsion (Ideal)

STEP 1

Select the desired working fluid or go with the default working fluid.

STEP 2

Enter the desired compressor inlet temperature value or go with the default temperature value.

STEP 3

Enter the desired compressor inlet pressure value or go with the default pressure value.

STEP 4

Enter the desired turbine inlet temperature value or go with the default temperature value.

STEP 5

Enter the desired turbine inlet pressure value or go with the default turbine inlet pressure value.

STEP 6

Enter the desired mass flow rate value or go with the default mass flow rate value.

STEP 7

Enter the desired fuel HHV value or go with the default fuel HHV value.

STEP 8

Click on the Calculate button to start the calculation of the Brayton cycle output values for the chosen input values.

STEP 9

When done with Steps 1 through 8, click on the Exit button to go back to the Main menu.

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Power Cycles: Rankine

This subsection provides analysis of the Rankine cycle.

Input Values:

Turbine Inlet Conditions (Temperature and Pressure), Steam Mass Flow Rate, Fuel HHV

Output Values:

Power Output, Fuel Consumption, Cycle Efficiency, Heat Rate

Assumptions:

Isentropic compression and expansion. Ideal combustion and heat transfer.

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Running - Power Cycles: Rankine

STEP 1

Select the desired steam turbine inlet conditions (temperature and pressure values) or go with the default selection.

STEP 2

Enter the steam mass flow rate value or go with the default steam mass flow rate value.

STEP 3

Enter the desired fuel HHV value or go with the default fuel HHV value.

STEP 4

Click on the Calculate button to start the calculation of the Rankine cycle output values for the chosen input values.

STEP 5

When done with Steps 1 through 4, click on the Exit button to go back to the Main menu.

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Power Cycle Components/Processes

This section provides analysis of power cycle components/processes (compression, combustion and expansion).

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Power Cycle Components/Processes: Compression

This subsection provides analysis of compression.

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Power Cycle Components/Processes: Compression: Isentropic (Ideal)

This subsection provides analysis of isentropic compression.

Input Values:

Working Fluid, Working Fluid Mass Flow Rate, Inlet Temperature, Inlet Pressure, Outlet Pressure

Output Values:

Power Input, Outlet Temperature

Assumptions:

Isentropic compression. Specific heat is constant.

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Running - Power Cycle Components/Processes: Compression: Isentropic (Ideal)

STEP 1

Select the desired working fluid or go with the default working fluid.

STEP 2

Enter the desired compressor inlet temperature value or go with the default temperature value.

STEP 3

Enter the desired compressor inlet pressure value or go with the default pressure value.

STEP 4

Enter the desired compressor outlet pressure value or go with the default pressure value.

STEP 5

Enter the desired compressor mass flow rate value or go with the default mass flow rate value.

STEP 6

Click on the Calculate button to start the calculation of compressor outlet temperature and power input values for the chosen input values.

STEP 7

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Power Cycle Components/Processes: Combustion

This subsection provides analysis of combustion.

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Power Cycle Components/Processes: Combustion: Coal/Oil

This subsection provides analysis of the combustion process when coal or oil are considered as the fuel.

Input Values:

Fuel Composition, Fuel Temperature, Oxidant Composition, Oxidant Temperature, Stoichiometry

Output Values:

Fuel HHV, Fuel Enthalpy, Oxidant Enthalpy, Oxidant to Fuel Ratio, Flame Temperature, Combustion Gas Composition

Assumptions:

Complete combustion. No gas dissociation. No heat loss.

Specific heat is not constant.

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Running - Power Cycle Components/Processes: Combustion: Coal/Oil

STEP 1

Enter the desired fuel (coal or oil) composition or go with the default fuel composition.

Note: If you decide to change the fuel composition, click on the Normalize button to normalize the fuel composition for you.

STEP 2

Enter the desired oxidant composition or go with the default oxidant composition.

Note: If you decide to change the oxidant composition, click on the Normalize button to normalize the oxidant composition for you.

Furthermore, click on the fuel Normalize button one more time to calculate the new oxidant to fuel ratio value for you.

STEP 3

Enter the desired fuel temperature value or go with the default temperature value.

STEP 4

Enter the desired oxidant temperature (preheat) value or go with the default temperature value.

STEP 5

Enter the desired stoichiometry value (1 or > 1) or go with the default stoichiometry value.

STEP 6

Click on the Calculate button to start the calculation of combustion gas composition and flame temperature value for the chosen input values.

STEP 7

When done with Steps 1 through 6, click on the Exit button to go back to the Main menu.

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Power Cycle Components/Processes: Combustion: Gas

This subsection provides analysis of the combustion process when gas is considered as the fuel.

Input Values:

Fuel Composition, Fuel Temperature, Oxidant Composition, Oxidant Temperature, Stoichiometry

Output Values:

Fuel HHV, Fuel Enthalpy, Oxidant Enthalpy, Oxidant to Fuel Ratio, Flame Temperature, Combustion Gas Composition

Assumptions:

Complete combustion. No gas dissociation. No heat loss.

Specific heat is not constant.

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Running - Power Cycle Components/Processes: Combustion: Gas

STEP 1

Enter the desired fuel (gas) composition or go with the default fuel composition.

Note: If you decide to change the fuel composition, click on the Normalize button to normalize the fuel composition for you.

STEP 2

Enter the desired oxidant composition or go with the default oxidant composition.

Note: If you decide to change the oxidant composition, click on the Normalize button to normalize the oxidant composition for you.

Furthermore, click on the fuel Normalize button one more time to calculate the new oxidant to fuel ratio value for you.

STEP 3

Enter the desired fuel temperature value or go with the default temperature value.

STEP 4

Enter the desired oxidant temperature (preheat) value or go with the default temperature value.

STEP 5

Enter the desired stoichiometry value (1 or > 1) or go with the default stoichiometry value.

STEP 6

Click on the Calculate button to start the calculation of combustion gas composition and flame temperature value for the chosen input values.

STEP 7

When done with Steps 1 through 6, click on the Exit button to go back to the Main menu.

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Power Cycle Components/Processes: Expansion

This subsection provides analysis of expansion.

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Power Cycle Components/Processes: Expansion: Isentropic (Ideal)

This subsection provides analysis of isentropic expansion.

Input Values:

Working Fluid, Working Fluid Mass Flow Rate, Inlet Temperature, Inlet Pressure, Outlet Pressure

Output Values:

Outlet Temperature, Power Output

Assumptions:

Isentropic expansion. Specific heat is constant.

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Running - Power Cycle Components/Processes: Expansion: Isentropic (Ideal)

STEP 1

Select the desired working fluid or go with the default working fluid.

STEP 2

Enter the desired turbine/expander inlet temperature value or go with the default temperature value.

STEP 3

Enter the desired turbine/expander inlet pressure value or go with the default pressure value.

STEP 4

Enter the desired turbine/expander outlet pressure value or go with the default pressure value.

STEP 5

Enter the desired turbine/expander mass flow rate value or go with the default mass flow rate value.

STEP 6

Click on the Calculate button to start the calculation of turbine/expander outlet temperature and power output values for the chosen input values.

STEP 7

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Compressible Flow

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Compressible Flow: Mach Number

This subsection provides analysis of Mach Number.

Input Values:

Working Fluid, Temperature, Velocity

Output Values:

Mach Number

Assumptions:

Specific heat is constant

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Running - Compressible Flow: Mach Number

STEP 1

Select the desired working fluid or go with the working fluid.

STEP 2

Enter the desired temperature value or go with the default temperature value.

STEP 3

Enter the desired velocity value or go with the default velocity value.

STEP 4

Click on the Calculate button to start the calculation of Mach number value for the chosen input values.

STEP 5

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Compressible Flow: Nozzle

This subsection provides analysis of nozzle.

Input Values:

Working Fluid, Stagnation Temperature, Stagnation Pressure, Velocity

Output Values:

Static Temperature, Static Pressure, Mach Number

Assumptions:

Specific heat is constant

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Running - Compressible Flow: Nozzle

STEP 1

Select the desired working fluid or go with the default working fluid.

STEP 2

Enter the desired stagnation temperature value or go with the default stagnation temperature value.

STEP 3

Enter the desired stagnation pressure value or go with the default stagnation pressure value.

STEP 4

Enter the desired velocity value or go with the default velocity value.

STEP 5

Click on the Calculate button to start the calculation of Mach number, stagnation temperature and stagnation pressure values for the chosen input values.

STEP 6

When done with Steps 1 through 5, click on the Exit button to go back to the Main menu.

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Compressible Flow: Normal Shock

This subsection provides analysis of normal shock.

Input Values:

Working Fluid, Inlet Stagnation Temperature, Inlet Stagnation Pressure, Inlet Velocity

Output Values:

Inlet Static Temperature, Inlet Static Pressure, Inlet Mach Number
Outlet Stagnation Temperature, Outlet Stagnation Pressure, Outlet Velocity, Outlet Static Temperature, Outlet Static Pressure, Outlet Mach Number

Assumptions:

Specific heat is constant

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Running - Compressible Flow: Normal Shock

STEP 1

Select the desired working fluid or go with the default working fluid.

STEP 2

Enter the desired inlet stagnation temperature value or go with the default inlet stagnation temperature value.

STEP 3

Enter the desired inlet stagnation pressure value or go with the default inlet stagnation pressure value.

STEP 4

Enter the desired inlet velocity value or go with the default velocity value.

STEP 5

Click on the Calculate button to start the calculation of inlet Mach number, inlet static temperature, inlet static pressure, outlet Mach number, outlet velocity, outlet static temperature, and outlet static pressure values for the chosen input values.

STEP 6

When done with Steps 1 through 5, click on the Exit button to go back to the Main menu.

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Compressible Flow: Diffuser

This subsection provides analysis of diffuser.

Input Values:

Working Fluid, Static Temperature, Static Pressure, Velocity

Output Values:

Mach Number, Stagnation Temperature, Stagnation Pressure

Assumptions:

Specific heat is constant

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Running - Compressible Flow: Diffuser

STEP 1

Select the desired working fluid or go with the default working fluid.

STEP 2

Enter the desired static temperature value or go with the default static temperature value.

STEP 3

Enter the desired static pressure value or go with the default static pressure value.

STEP 4

Enter the desired velocity value or go with the default velocity value.

STEP 5

Click on the Calculate button to start the calculation of Mach number, stagnation temperature and stagnation pressure values for the chosen input values.

STEP 6

When done with Steps 1 through 5, click on the Exit button to go back to the Main menu.

Engineering Software

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Compressible Flow: Thrust

This subsection provides analysis of thrust.

Input Values:

Working Fluid, Working Fluid Mass Flow Rate, Stagnation Temperature,
Stagnation Pressure, Velocity, Ambient Pressure

Output Values:

Static Temperature, Static Pressure, Mach Number, Thrust

Assumptions:

Specific heat is constant

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Running - Compressible Flow: Thrust

STEP 1

Select the desired working fluid or go with the default working fluid.

STEP 2

Enter the desired stagnation temperature value or go with the default stagnation temperature value.

STEP 3

Enter the desired stagnation pressure value or go with the default stagnation pressure value.

STEP 4

Enter the desired velocity value or go with the default velocity value.

STEP 5

Enter the desired mass flow rate value or go with the default mass flow rate value.

STEP 6

Enter the desired ambient pressure value or go with the default ambient pressure value.

STEP 7

Click on the Calculate button to start the calculation of Mach number, static temperature, static pressure and thrust values for the chosen input values.

STEP 8

When done with Steps 1 through 7, click on the Exit button to go back to the Main menu.